



## Original Article

Health monitoring of carbon fiber-reinforced polymer composites in  $\gamma$ -radiation environment using embedded fiber Bragg grating sensorsJing Zhong<sup>a</sup>, Feida Chen<sup>a, c, \*</sup>, Yuehao Rui<sup>a</sup>, Yong Li<sup>b</sup>, Xiaobin Tang<sup>a, c, \*</sup><sup>a</sup> Department of Nuclear Science and Technology, Nanjing University of Aeronautics and Astronautics, Nanjing, 211106, China<sup>b</sup> College of Materials Science and Technology, Nanjing University of Aeronautics and Astronautics, Nanjing, 211106, China<sup>c</sup> Key Laboratory of Nuclear Technology Application and Radiation Protection in Astronautics, Ministry of Industry and Information Technology, Nanjing, 211106, China

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## ABSTRACT

Fiber-reinforced polymer (FRP) composites are considered suitable candidates for structural materials of spacecrafts due to their excellent properties of high strength, light weight, and corrosion resistance. An online health monitoring method for FRP composites must be applied to space structures. However, the application of existing health monitoring methods to space structures is limited due to the harsh space environment. Here, carbon fiber-reinforced polymer (CFRP) composites embedded with fiber Bragg grating (FBG) sensors were prepared to explore the feasibility of strain monitoring using embedded FBG sensors in  $\gamma$ -radiation environment. The analysis of the influence of radiation on the strain monitoring demonstrated that the embedded FBG can be successfully applied to the health monitoring of FRP composites in radiation environment.

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## 1. Introduction

Fiber-reinforced polymer (FRP) composites are commonly used in the aerospace, automotive, and construction industries due to their superior properties of light weight, high strength, corrosion endurance, fatigue resistance, and strong designability [1–5]. Hence, FRP composites demonstrate high potential as structural materials for spacecrafts. However, the space environment where spacecrafts operate can significantly degrade the properties of FRP composites because of its extremely harsh characteristics of high vacuum, thermal cycles, atomic oxygen, ultraviolet light, radiation, and presence of space debris [6–8]. FRP composites exposed to the space environment suffer from strength change, interface debonding, surface cracking, brittle fracture, molecular chain breakage, and other types of damage [9–16]. Thus, an effective health monitoring method for FRP composites is crucial to ensure their long-term reliability in service.

Numerous health monitoring methods for FRP composites, such as ultrasonic detection radiographic inspection, acoustic emission

monitoring, strain gauge testing, and optical fiber detection, have been proposed [17–25]. Although ultrasonic and radiographic inspection methods are mature and widely used, they can only be utilized for offline testing. Frequent debonding of the interface between piezoelectric sensors and composites in acoustic emission inspections introduces noise during the propagation process of elastic waves in materials. Strain gauge sensors are easily embedded in composite materials, but their electrical signals are susceptible to electromagnetic interference. By contrast, optical fiber Bragg grating (FBG) sensors have attracted considerable research attention in the field of damage detection due to their embeddability, anti-electromagnetic interference capabilities and characteristics of wavelength encoded. Previous studies verified the feasibility of the embedded FBG sensor technique in dynamic stress monitoring, thermal cycle detection, and thermal deformation measurement [26–30]. However, the harsh environment may also exert some effect on the monitoring method itself. When FBG sensors are applied to radiation environment for sensing, it was found that high-energy ionizing radiation can cause signals drifting and transmission loss to FBGs, which can introduce errors for the sensing of temperature or strain [31–33]. On the basis, relevant studies have focused on improving the radiation resistance of FBGs mainly by fiber composition modulation, grating inscription techniques and pre-processing method to mitigate the effects of

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radiation on FBGs [34–36]. However, it is still unknown whether FBG sensors are capable for materials monitoring in radiation environments.

In this work, FBG sensors were embedded in FRP composites for strain monitoring of materials in  $\gamma$ -radiation environment. The quantitative model of the relationship between FBG wavelength shift and FRP strain was established, and the relationship between FBG wavelength and temperature was determined. Moreover, the irradiation experiment proves that the compatibility between FBG and FRP materials maintains well after irradiation and the coefficients of strain sensitivity of FBG sensors embedded in FRP composites are slightly affected by the radiation. Hence, the embedded FBG is very suitable for materials monitoring in radiation environments.

## 2. Working principle of FBG

FBG sensor is an optical fiber with a regular distribution of grating. The grating is inscribed inside the fiber core via exposure to ultraviolet light or femtosecond laser. The refractive index of optical fiber varies periodically at a certain length. Notably, FBG is a kind of superior bandpass filter that can reflect a narrow spectrum of incident light and transmit the rest at the same time [37]. The structure schematic is shown in Fig. 1. Bragg wavelength ( $\lambda_B$ ) is the central wavelength in the reflection spectrum of FBG and defined as follows:

$$\lambda_B = 2n_e \Lambda, \tag{1}$$

where  $n_e$  is the effective refractive index of the optical fiber core and  $\Lambda$  is the period of Bragg grating. Temperature and strain can affect  $n_e$  and  $\Lambda$  and further lead to shifting in  $\lambda_B$ . The wavelength shift subjected to strain and temperature is expressed as follows:

$$\Delta\lambda_B = K_\epsilon \Delta\epsilon + K_T \Delta T, \tag{2}$$

where  $\Delta\lambda_B$ ,  $\Delta\epsilon$ , and  $\Delta T$  represent the shift of  $\lambda_B$ , strain, and temperature, respectively;  $K_\epsilon$  and  $K_T$  are coefficients of strain and temperature sensitivity, respectively, which are related to parameters of fiber and grating [38]. The wavelength shift is linearly proportional to the variation in temperature and strain to ensure that strain and temperature can be monitored by FBG.

## 3. Experiments

### 3.1. Selection of FBG sensors

For the reliable strategy for monitoring and sensing in radiation environments, FBG sensors with resistance to radiation should be selected properly. Currently, the FBGs written on pure silica fibers or F-doped fibers are considered to be of low radiation sensitivity, which present lower radiation loss and wavelength shift than that written on the standard Ge-doped fibers [39–41]. Additionally, FBGs fabricated with femtosecond laser developed in recent years, compared to the conventional UV laser inscription technique, show better high-temperature stability and better resistance to radiation-induced wavelength shift [42–45]. Here, the resistance to radiation and commercial access of various types of FBG sensors have been taken into comprehensive consideration. In this study, FBG sensors inscribed onto pure silica fiber SM1250SC(9/125)P (FemtoFiberTec, Germany) with good resistance to radiation were selected. The FBGs were written by femtosecond laser point-by-point writing technique and with a polyimide coating and a grating length of 4 mm.

### 3.2. Specimen manufacturing

Carbon fiber-reinforced polymer (CFRP) laminates in this study were manufactured using the mold pressing technique with unidirectional carbon fiber/epoxy prepreg USN15000/7901/RC33(Guangwei Composites Co., China). The dimensions of specimen were 90 mm  $\times$  13 mm  $\times$  2 mm (length  $\times$  width  $\times$  thickness) according to the ASTM D7264 standard. The stacking sequence was [0<sub>4</sub>/90<sub>4</sub>/0<sub>4</sub>/90<sub>4</sub>/0<sub>4</sub>]. FBG sensors were located between the third and fourth prepreg layers along the direction of the carbon fiber, that is, [0<sub>3</sub>/FBG/0/90<sub>4</sub>/0<sub>4</sub>/90<sub>4</sub>/0<sub>4</sub>]. The ends of FBG sensors were taken out from the side of the laminate, inserted into the Teflon tube, and sealed with silicone rubber to prevent the optical fiber from breaking due to the curing of the resin. The schematic of the FBG embedded in CFRP composites is shown in Fig. 2. The temperature in the curing process was raised from room temperature to 80 °C at 1 °C/min, maintained for 0.5 h, increased to 120 °C at 1 °C/min, and held for 1.5 h with a pressure of 12 MPa. Finally, the laminates were cooled to room temperature naturally.

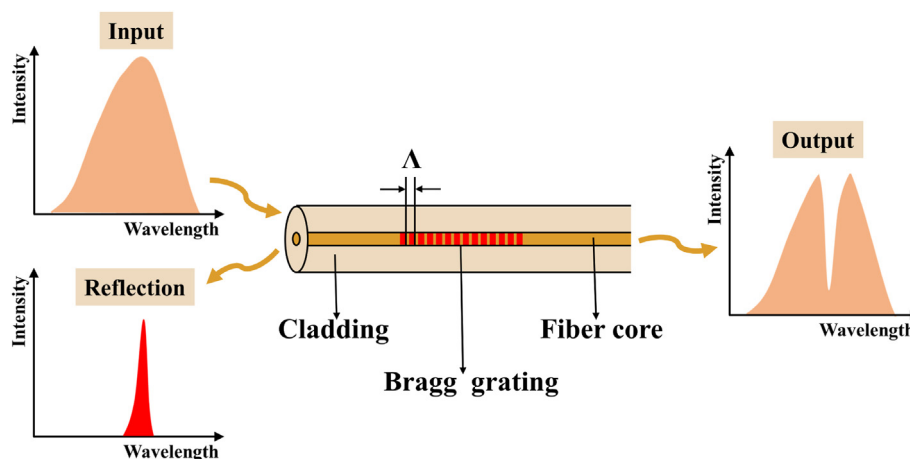


Fig. 1. Structure schematic of the FBG sensor consisting of cladding, fiber core, and Bragg grating. The regular distribution of grating allows the reflection of a narrow spectrum of incident light.

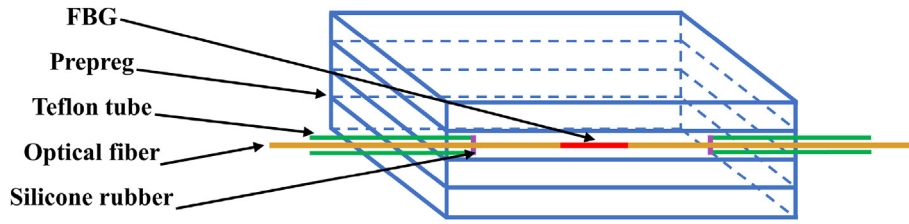


Fig. 2. Schematic of the FBG embedded in CFRP composites.

3.3. Measurement of the coefficient of strain sensitivity

Coefficients of strain sensitivity were detected with a mechanical testing machine WDW-100 (WANCE, China) under three-point flexural loading. The reflected signals of FBG sensors were recorded using an FBG demodulator SuperHawk 6000 (Hope-Excellence Information Technology Co., China). As shown in Fig. 3, the specimen embedded with FBG was secured on the fixture with the margin of FBG connected to the wavelength demodulator, and the demodulator was connected to a laptop. The loading speed was controlled at 1 mm/min.

3.4. Effects of temperature on FBG sensors

FBG sensors were placed in a high-low-temperature test chamber for measuring their response to the temperature range of 0 °C–50 °C. Coefficients of temperature sensitivity of FBG before and after being embedded in CFRP were measured. The initial temperature was set to 50 °C, and the next temperature decreased in gradients of 10 °C. The temperature was stabilized for 30 min after each adjustment, and wavelength data were recorded.

3.5.  $\gamma$ -radiation experiments

A  $^{60}\text{Co}$   $\gamma$ -radiation facility (Nanjing University of Aeronautics and Astronautics Irradiation Center) was employed as the radiation resource. Embedded FBGs and free FBGs are presented in Fig. 4 and Table 1, where “E” and “F” represent embedded FBGs and free FBGs, respectively. All the grating of free and embedded FBGs are not bare, which are covered with a polyimide coating. Free FBGs were attached to the acrylic plate with polyimide tape to prevent wavelength interference due to optical fiber bending. The irradiation time were 10, 20 30, 49, 69, and 98 h. The irradiation dose rate

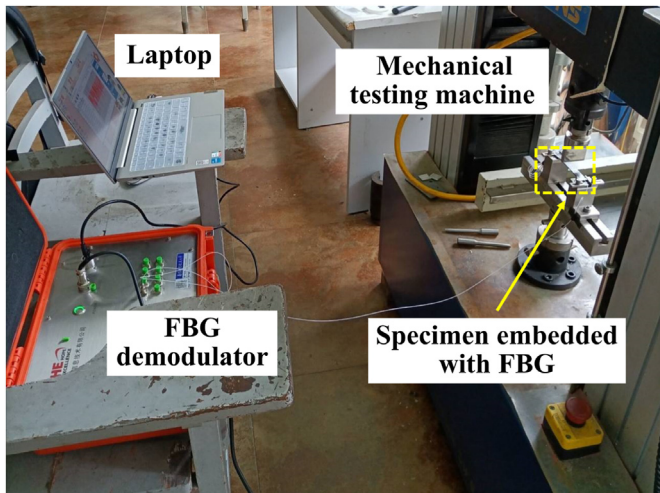


Fig. 3. Experimental setup for testing the coefficient of strain sensitivity.

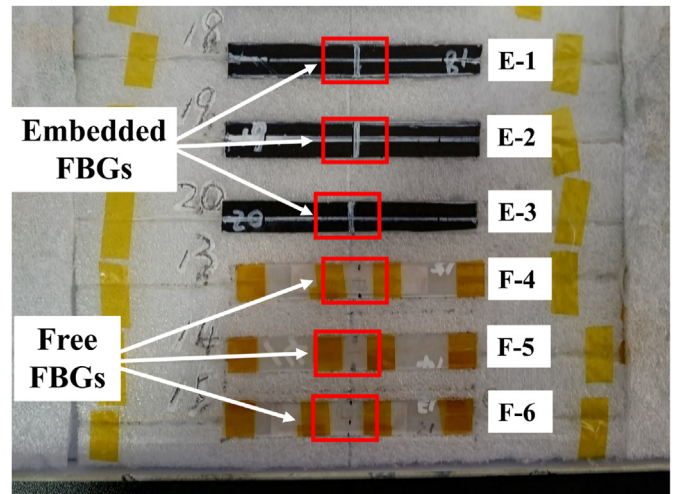


Fig. 4. Photos of embedded FBGs and free FBGs (“E” and “F” represent embedded FBGs and free FBGs, respectively).

Table 1

FBG sensors in this work (“E” and “F” represent embedded FBGs and free FBGs, respectively).

FBG label	1	2	3	4	5	6
Embedded/Free	E	E	E	F	F	F

was 10.19 kGy/h, and irradiation doses were 101.9, 203.8, 305.7, 499.3, 703.1, and 998.6 kGy, which were calibrated with a silver dichromate dosimeter. Wavelength data of FBG sensors were recorded at 30 °C after each dose of irradiation.

4. Results and discussion

4.1. Coefficients of strain sensitivity of FBG sensors

The linear relationship between wavelength shift and the applied load is illustrated in Fig. 5a, in which data are normalized to the unloaded wavelength. The linear fitting curves of the wavelength shift versus applied strain are shown in Fig. 5b, which refers to the strain of composites rather than FBG sensors. Slopes of curves, that is coefficients of strain sensitivity ( $K_\epsilon$ ), are listed in Table 2. The mechanical deformation and photoelastic effect result in the wavelength shift of FBG under load. In addition, the linear response of wavelength to strain reflects the satisfactory compatibility of embedded FBGs and composites. Curves of wavelength versus strain will change abruptly because the strain that transfers to FBG becomes uneven if FBG is delaminated with composites.

4.2. Effects of temperature on FBG sensors

The wavelength shift of free FBGs with temperature is shown in

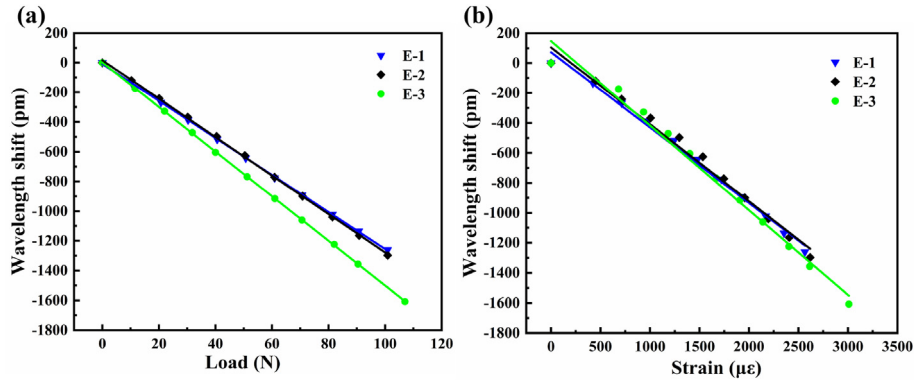


Fig. 5. Response of FBG sensors with applied strain. Relationship between the (a) wavelength of FBGs embedded in the CFRP and load and (b) wavelength shift of embedded FBGs and applied strain (“E” represents the embedded FBG).

Table 2  
Coefficients of strain sensitivity for embedded FBGs (“E” represents embedded FBG).

FBG label	E-1	E-2	E-3
$K_\epsilon$ (pm/ $\mu\epsilon$ )	-0.503	-0.513	-0.564

Fig. 6a, in which data are normalized to the wavelength at 0 °C. The wavelength shift of FBGs demonstrated a positive linear relationship with temperature due to thermal expansion and thermo-optic effect. Besides, thermal deformation of CFRP also affects the wavelength of FBGs when FBGs are embedded in composites. The response of FBG to temperature before and after being embedded in CFRP is displayed in Fig. 6b.  $K_T$  values for each FBG used in this work are listed in Table 3. According to the slope of the fitting curve, the coefficient of temperature sensitivity of FBG increases after being embedded in CFRP because the thermal expansion of CFRP leads to the additional strain exerted on FBG. Furthermore, the wavelength shift attributed to the thermal strain of CFRP was obtained when the wavelength shift of the free FBG was subtracted from that of the embedded FBG. The thermal strain of CFRP is insignificant in the temperature range of 0 °C–50 °C. The interference of temperature on strain measurement via FBG can be corrected on the basis of the linear relationship between the temperature and FBG wavelength.

Table 3  
Coefficients of temperature sensitivity of FBGs before and after being embedded in the CFRP in this work.

FBG label	E-1	E-2	E-3	F-4	F-5	F-6
$K_T$ before being embedded (pm/°C)	10.2	10.4	10.1	10.4	9.9	10.2
$K_T$ after being embedded (pm/°C)	12.8	11.4	11.8	–	–	–

4.3. Effects of radiation on FBG sensors

The curves of wavelength shift versus temperature and applied strain after a  $\gamma$ -irradiation dose of 998.6 kGy are depicted in Fig. 7 to investigate the effects of radiation on FBG sensors. As illustrated in Fig. 7a, the wavelength shift is still linearly proportional to the applied strain. Therefore, the FBG sensors remain well bonded to the materials after irradiation.  $K_\epsilon$  values of embedded FBGs after irradiation are listed in Table 4. Although  $K_\epsilon$  of each FBG varies slightly after a  $\gamma$ -irradiation dose of 998.6 kGy, the value is negligible and within the experimentally allowed error range. Thus, the FBG sensor is suitable for internal strain sensing of CFRP in radiation environments. The curves of wavelength shift versus temperature are shown in Fig. 7b.  $K_T$  values of free and embedded FBGs after a  $\gamma$ -irradiation dose of 998.6 kGy are listed in Table 5.  $K_T$  of each FBG also changes slightly after irradiation.

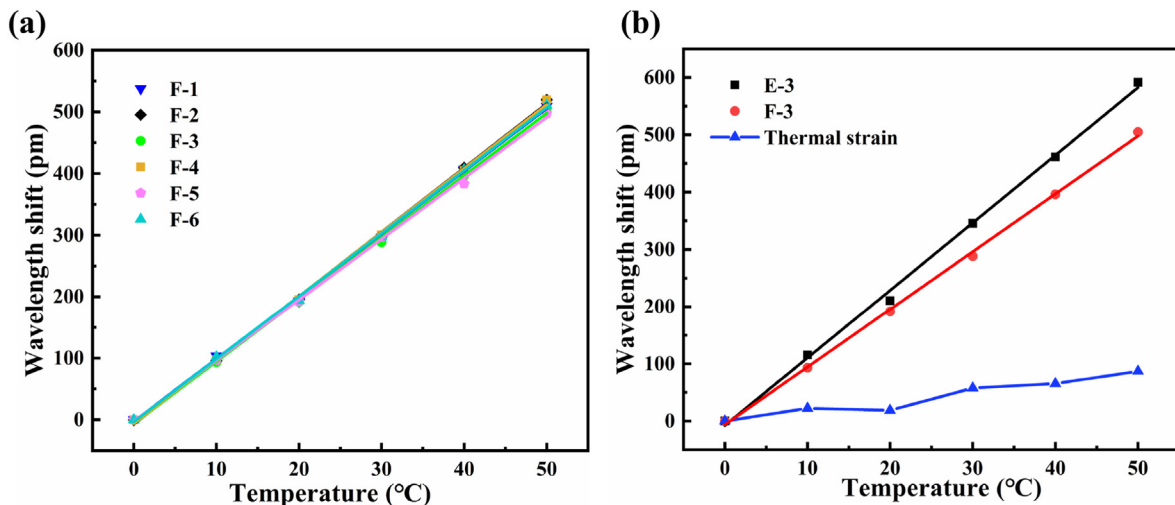


Fig. 6. Response of FBG sensors to ambient temperature. (a) Relationship between wavelength shift of free FBGs and temperature and (b) comparison of temperature sensitivity measured by the FBG before and after being embedded in CFRP (“E” and “F” represent embedded FBG and free FBG, respectively).

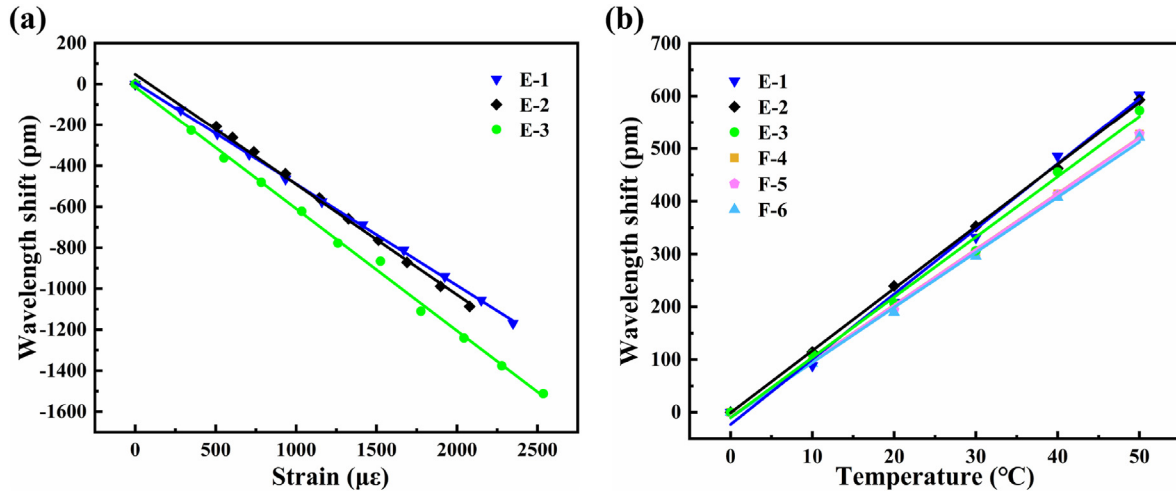


Fig. 7. Relationship between the wavelength of FBG and strain and temperature after a  $\gamma$ -irradiation dose of 998.6 kGy. (a) Applied strain and (b) temperature (“E” and “F” represent embedded FBG and free FBG, respectively).

**Table 4**  
Coefficients of strain sensitivity for embedded FBGs after a  $\gamma$ -irradiation dose of 998.6 kGy (“E” represents embedded FBG).

FBG label	E-1	E-2	E-3
$K_e$ after irradiation (pm/ $\mu\epsilon$ )	-0.495	-0.539	-0.596

**Table 5**  
Coefficients of temperature sensitivity for free and embedded FBGs after a  $\gamma$ -irradiation dose of 998.6 kGy (“E” and “F” represent embedded FBGs and free FBGs, respectively).

FBG label	E-1	E-2	E-3	F-4	F-5	F-6
$K_T$ after irradiation (pm/ $^{\circ}C$ )	12.4	11.8	11.4	10.8	10.6	10.5

4.4. Practical application in the monitoring of radiation-induced strain

The wavelength shift of FBGs is presented in Fig. 8a, including three embedded FBGs and three free FBGs, in the  $\gamma$ -radiation dose range of 0–998.6 kGy. The wavelength of three embedded FBGs

raises as the irradiation dose increases. Meanwhile, the wavelength shift of three free FBGs after irradiation was also confirmed, given that the refractive index of the quartz material is influenced by high-energy rays. This phenomenon will lead to errors in the monitoring of the FBG wavelength. This work considered that the effect of radiation on the wavelength of embedded FBG lies in two aspects. On one hand, the direct effect of  $\gamma$ -radiation on the FBG itself leads to a wavelength shift. On the other hand, the internal strain of composites develops after irradiation, thereby altering the wavelength of FBG.

In this study, the free FBGs were set as a control group to test the wavelength shift caused by the direct effect of  $\gamma$  radiation on FBGs. To eliminate the direct effect of radiation on FBG sensors, the average wavelength shift of three free FBGs was subtracted from the wavelength shift of each embedded FBG, thus the wavelength shift of FBG induced by the internal strain of composites was obtained, as shown in Fig. S1. The internal strain of composites induced by  $\gamma$  irradiation was further obtained according to Eq. (2) and  $K_e$  value in Table 4. The radiation-induced strain versus irradiation dose for three embedded FBGs is presented in Fig. 8b. The radiation-induced strain increases with the increase of irradiation,

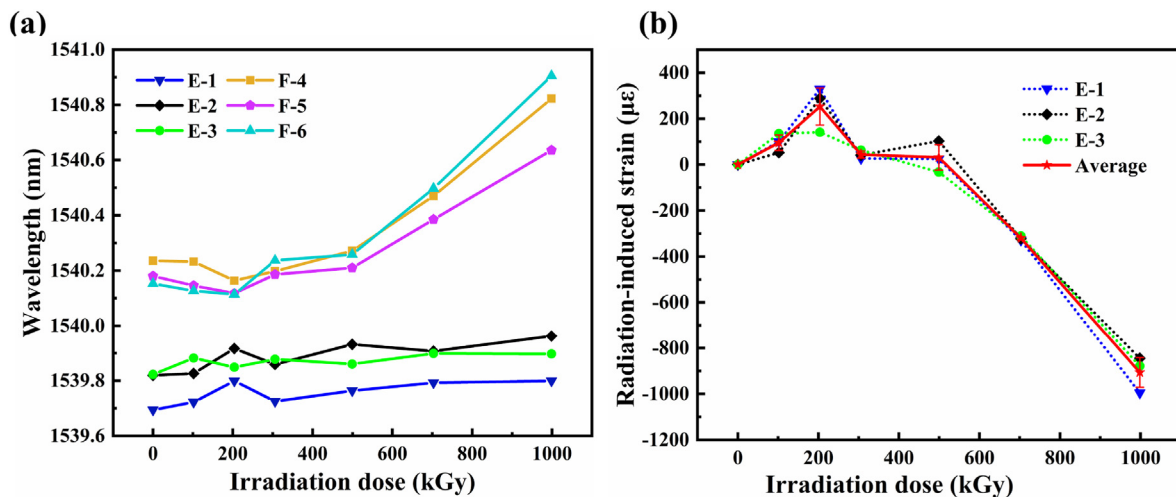


Fig. 8. Monitored radiation-induced strain of CFRP in the  $\gamma$ -irradiation dose of 0–988.6 kGy. (a) Wavelength monitored by free and embedded FBGs and (b) radiation-induced strain of CFRP monitored by embedded FBGs (“E” and “F” represent embedded FBG and free FBG, respectively).

which reaches a maximum when the irradiation dose is 203.8 kGy. However, the radiation-induced strain decreases with the continuous increase of irradiation dose. The radiation-induced strain becomes a negative value and further increases in reverse as the irradiation dose increases until it reaches 499.3 kGy. The average radiation-induced strain of CFRP monitored by FBG is 906.2  $\mu\epsilon$  after exposure to an irradiation dose of 998.6 kGy.

## 5. Conclusion

A monitoring technique based on embedded FBG was developed in this study to assess the health status of CFRP composites in  $\gamma$ -radiation environment. The quantitative model of the relationship between the FBG wavelength shift and CFRP strain was established. The relationship between the temperature and FBG wavelength shift was determined to correct the influence of temperature on strain measurement. Furthermore, this work demonstrated the negligible effect of radiation on coefficients of strain sensitivity and verified the applicability of the FBG method in radiation environment. The relationship of material strain monitored by FBG versus the radiation dose was acquired in the irradiation dose range of 0–988.62 kGy. In summary, strain detection based on FBG is an effective health monitoring method for evaluating CFRP composites exposed to radiation environment. This work provides a feasible strategy for the nondestructive evaluation of health status of CFRP composites in radiation environment. Meanwhile, a future investigation can focus on determining whether or not the effect of high-dose radiation on the coefficient of strain sensitivity of FBG is negligible. In addition, with the development of new fabrication techniques, it is expected that FBG sensors with excellent resistance to radiation will be commercially available in the near future. And their low radiation-induced wavelength shift will contribute to superiority for strain sensing in radiation environment.

## Declaration of competing interest

The authors declare that they have no known competing financial interests or personal relationships that could have appeared to influence the work reported in this paper.

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## Appendix A. Supplementary data

Supplementary data to this article can be found online at <https://doi.org/10.1016/j.net.2023.04.028>.

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